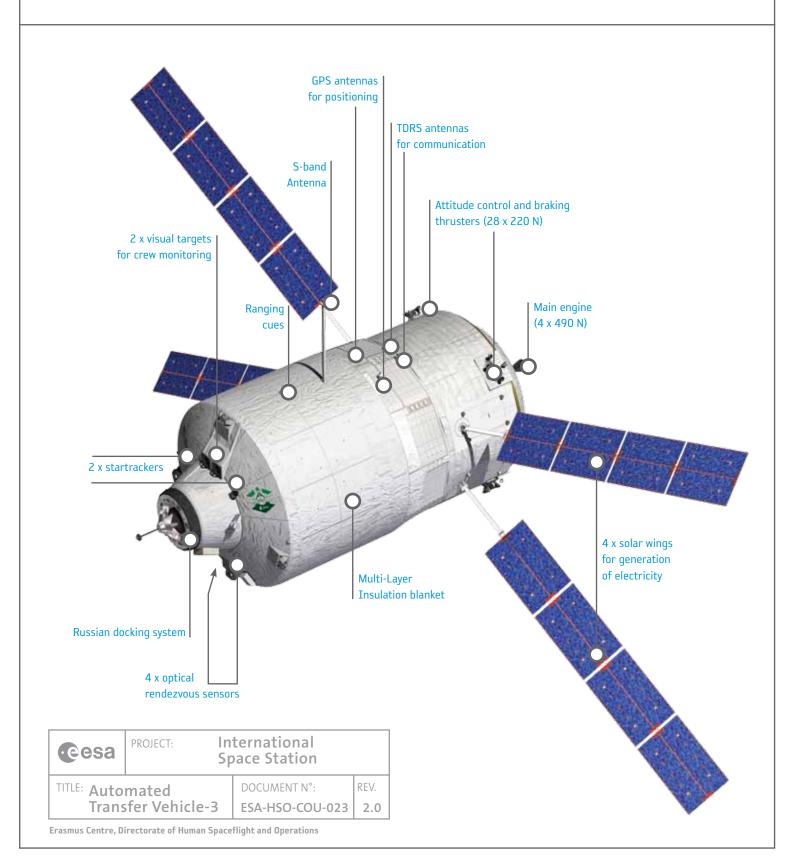
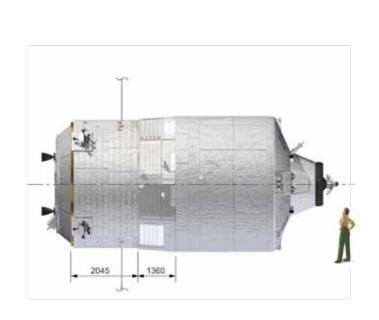
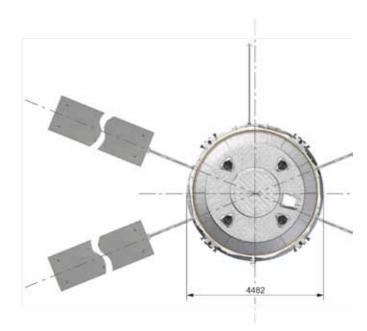
## → ATV EDOARDO AMALDI

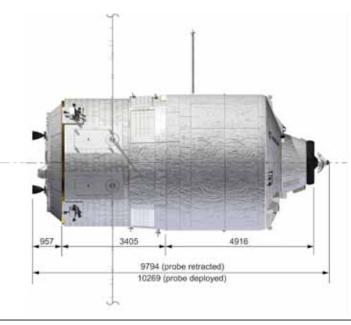
## European servicing and logistics vehicle

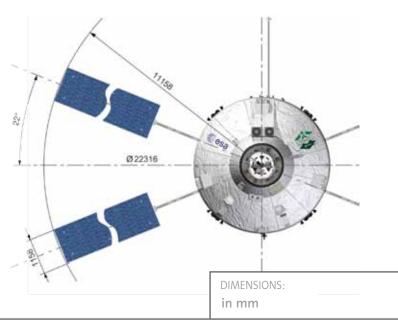
The Automated Transfer Vehicle (ATV) is an unmanned automatic vehicle sent into orbit by the European Ariane 5 launcher. It provides the International Space Station with: cargo, water, air, nitrogen, oxygen and attitude control propellant. It also removes waste from the Station and reboosts it to a higher altitude to compensate for atmospheric drag.















Specifications							
DIMENSIONS		THERMAL/ENVIRONMENTAL CONTROL					
Length: Largest diameter: Solar array span:	9794 mm (probe retracted) 4480 mm 22 281 mm	Thermal Control:		Multi-Layer Insulation material, active thermal control using variable & constant conductive heat pipes and paints			
MASS BUDGET Vehicle dry mass:	Vehicle dry mass: 9778 kg		ECLSS:		Fire detection, air circulation, air temperature monitoring		
Vehicle consumables: Total vehicle mass:	2261 kg 12 039 kg	ELECTRICAL POWER					
Total cargo upload capacity: Mass at launch (max):	7384 kg 20 100 kg	Ascent to ISS and deorbit:		and 40	4 solar wings of 4 panels each and 40 Ah rechargeable batteries		
Waste download capacity:	6495 kg (420 km altitude, 51.6° inclination)	Number of arra Number of pan Generated pow	els/array:	4 4 3800 \	N after 6 months in o	rhit	
PROPULSION		Required power:		< 400 W dormant mode			
Main propulsion system:	4 x 490 N thrusters (pressurised liquid bipropellant system)	Supplied by ISS:     < 900 W active mode   MAIN CONSTRUCTION MATERIAL					
Attitude control system:	28 x 220 N thrusters (pressurised liquid bipropellant system)	Pressure shell: Micrometeoroid and		Al-2219 Primary bumper: Al-6061-T6			
Propellant:	Monomethyl hydrazine fuel and Nitrogen tetroxide oxidiser	Debris Protection System: Internal structure (racks):		Secondary bumper: Nextel/Kevlar blankets Al-6061-T6			
Pressurisation: Helium at 31 MPa		Thermal insulation:		Goldised Kapton Multi-Layer Insulation blanket & aluminised beta cloth			
<ul> <li>AVIONICS</li> <li>2-failure-tolerant architecture</li> <li>Equipment interconnection via multiple redundant MIL-STD-1553B buses</li> <li>2-fault-tolerant computer via voting mechanisms</li> <li>Flight Application Software: 450 000 lines of code</li> </ul>		Solar arrays: MAIN CONTRACTOR		Silicon solar cells on 4 carbon fibre reinforced plastic sandwich panels			
		Astrium-Space Transportation, leading a consortium of many subcontractors					
COMMUNICATIONS INFRAST To ground: ATV to ISS:	RUCTURE S-band via TDRS satellite S-band antenna via proximity link	-					
<ul> <li>RELATIVE NAVIGATION</li> <li>Relative GPS</li> <li>Optical rendezvous sensors</li> </ul>		eesa	PROJECT: International Space Station				
		TITLE: Autor Trans	nated fer Vehic	le-3	DOCUMENT N°: ESA-HSO-COU-023	REV. 2.0	





ATV Johannes Kepler approach photographed by ESA astronaut Paolo Nespoli on the International Space Station

## Utilisation Data

LAUNCH CONFIGURATION		IN-ORBIT CONFIGURATION
Payload: Envelope: Cargo mass:	8 racks with 1.25 m <sup>3</sup> each 1.005 m <sup>3</sup> in front of 2 racks Dry cargo: 2200 kg Water: 285 kg	Deployed solar array, with a total span of 22.3 m, provide electrical power to rechargeable batteries for eclipse periods. Automated flight towards the International Space Station.
	Gas (oxygen and air): 100 kg ISS refuelling propellant: 860 kg (306 kg of fuel, 554 kg of oxidiser) ISS reboost and attitude control propellant: 3150 kg Total cargo upload capacity: 6595 kg	FLIGHT HARDWARE Propulsion and reboost system Avionics equipment Guidance navigation and control system Communications system
Launch vehicle:		Power generation and storage system Thermal control system Russian docking and refuelling system
Launch site:	Kourou, French Guiana	
Launch date:	9 March 2012	
	(as of 4 February 2012)	

